EE16B Designing Information Devices and Systems II

Lecture 8A Observability and Observers

Outputs

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 $\vec{x}(t+1) = A\vec{x}(t) + Bu(t)$

Can't always measure state directly or all states...

 $\vec{y}(t) = C\vec{x}(t)$

Define output:

p x n matrix for p outputs

Observability

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A system is "observable" if, by watching y(0),y(1),y(2),... we can determine the full state

Two stage approach:

1) Determine initial state x(0) from y(0), y(1), ...

2)
$$\vec{x}(t) = A^t \vec{x}(0) + Bu(t)$$

$$\vec{y} = \begin{bmatrix} y(0) \\ y(1) \\ \vdots \\ y(t) \end{bmatrix} = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^t \\ \triangleq O_t \end{bmatrix} \vec{x}(0)$$









Choosing L for Observer
$\vec{x}(t+1) = A\vec{x}(t) + Bu(t)$
$\hat{x}(t+1) = A\hat{x}(t) + Bu(t) + L(C\hat{x}(t) - y(t))$
$\vec{e}(t) \stackrel{\Delta}{=} \hat{x}(t) - \vec{x}(t)$ $\overrightarrow{C}\vec{x}(t)$
$\vec{e}(t+1) = \hat{x}(t+1) - \vec{x}(t+1)$
$= A\left(\hat{x}(t) - \vec{x}(t)\right) - LC(\hat{x}(t) - x(t))$
$\overline{ec{e}(t)}$ $ec{e}(t)$
$\vec{e}(t+1) = (A + LC)\vec{e}(t)$
$ec{e}(t) ightarrow 0$ If (A + LC) has eigenvalues inside unit circle



Back to Example
$\vec{x}(t+1) = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix} \vec{x}(t) \qquad y(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} \vec{x}(t)$
$\hat{x}(t+1) = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix} \hat{x}(t) + \begin{bmatrix} l_1 \\ l_2 \end{bmatrix} (\begin{bmatrix} 1 & 0 \end{bmatrix} \hat{x}(t) - y(t))$ eigenvalues of A+LC must be inside unit circle
$\theta = \frac{\pi}{2} \qquad \begin{bmatrix} 0 & -1 \\ 1 & 0 \end{bmatrix} + \begin{bmatrix} l_1 \\ l_2 \end{bmatrix} \begin{bmatrix} 1 & 0 \end{bmatrix} = \begin{bmatrix} l_1 & -1 \\ 1 + l_2 & 0 \end{bmatrix}$
$\lambda^{2} - l_{1}\lambda + (l_{2} + 1) = 0 \qquad \lambda_{1,2} = \pm 0.9i \qquad \Rightarrow \lambda^{2} + 0.81 = 0$ = 0 = 0.81
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Example
$\hat{x}(t+1) = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix} \hat{x}(t) + \begin{bmatrix} l_1 \\ l_2 \end{bmatrix} (\begin{bmatrix} 1 & 0 \end{bmatrix} \hat{x}(t) - y(t))$
$l_1 = 0 l_2 = -0.19 \vec{x}(0) = \begin{bmatrix} 1 \\ 1 \end{bmatrix} \hat{x}(0) = \begin{bmatrix} -1 \\ -1.3 \end{bmatrix}$
$ec{e}(t)=ec{x}(t)-\hat{x}(t)$



Kalman Filter

• We have not assumed noise and errors in our system model and inputs

$$\hat{x}(t+1) = A\hat{x}(t) + Bu(t) + L(C\hat{x}(t) - y(t))$$

Copy of system model

correction

A more elaborate form of the observer where the matrix L is also updated at each time, is known as the Kalman Filter and is the industry standard in navigation. The Kalman Filter takes into account the statistical properties of the noise that corrupts measurements and minimizes the mean square error between x(t) and $x^{*}(t)$



Figure 3: Rudolf Kalman (1930-2016 introduced the Kalman Filter as well as many of the state space concepts we studied, such as controllability an observability. He was awarded the National Medal of Science in 2009.

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Control Recap – State Feedback		
$u(t) = K \vec{x}(t)$		
Closed-loop system: $\Rightarrow ec{x}(t+1) = (A+BK)ec{x}(t)$		
Must choose K s.t. A+BK has eigenvalues inside the unit circle (or left half-plane for coninuous time)		
If controllable, can assign eigenvalues for A+BK arbitrarily		
If not, some eigenvalues of A can not be changed! (could be OK, if stable, bad news if not)		
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Control Recap	$\begin{bmatrix} y(0) \\ (1) \end{bmatrix} \begin{bmatrix} C \\ C \end{bmatrix}$
Observability:	$\begin{bmatrix} y(1) \\ \vdots \\ y(n-1) \end{bmatrix} = \begin{bmatrix} CA \\ \vdots \\ CA^{n-1} \end{bmatrix} \vec{x}(0)$
	O _{n-1} must have n independent rows (full rank) to determine x(0) uniquely from output
Duality: Observability of controllability of	(C,A) is the same as (A^{T},C^{T})
Guidance, Navigation & Co	ontrol (GNC) is aerospace engineering
Open loop Observers or "kalman" filters	eedback
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