

Lecture 14: Beam Bending

• Announcements:

- HW#4 online and due Tuesday, March 18
- Lecture Module 8 online
- Midterm is nearing: Thursday, March 20
 - ↳ I will soon pass out materials associated with the midterm, including an information sheet and old exams

• Reading: Senturia, Chpt. 8

• Lecture Topics:

- ↳ Stress, strain, etc., for isotropic materials
- ↳ Thin films: thermal stress, residual stress, and stress gradients
- ↳ Internal dissipation
- ↳ MEMS material properties and performance metrics

• Reading: Senturia, Chpt. 9

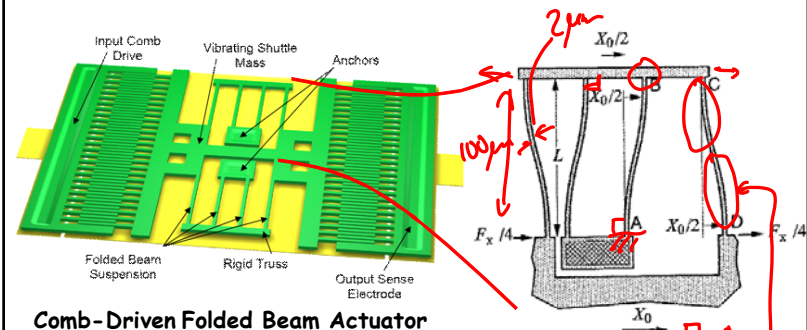
• Lecture Topics:

- ↳ Bending of beams
- ↳ Cantilever beam under small deflections
- ↳ Combining cantilevers in series and parallel
- ↳ Folded suspensions
- ↳ Design implications of residual stress and stress gradients

• Last Time:

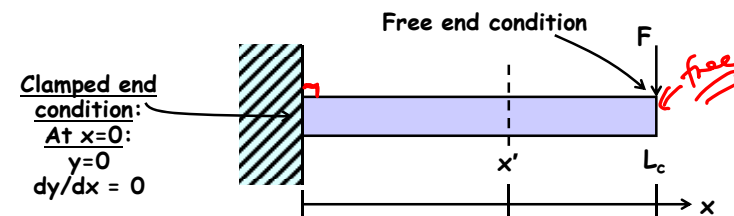
- Went through Module 7 on Mechanics of Materials
- Now finish this
- Then, start a new topic: Bending of Beams

- Springs and suspensions very common in MEMS
- Coils are popular in the macro-world; but not easy to make in the micro-world
- Beams: simpler to fabricate and analyze; become "stronger" on the micro-scale → use beams for MEMS

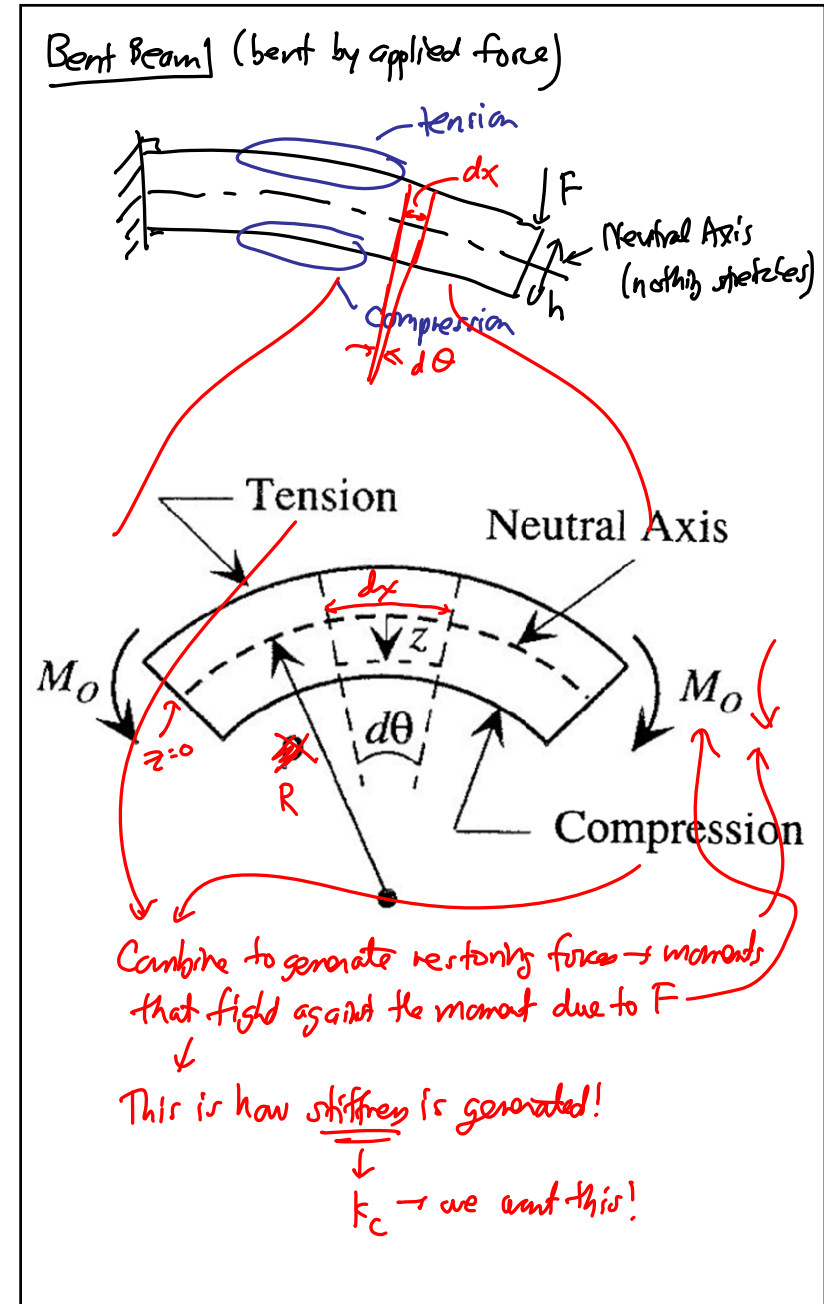
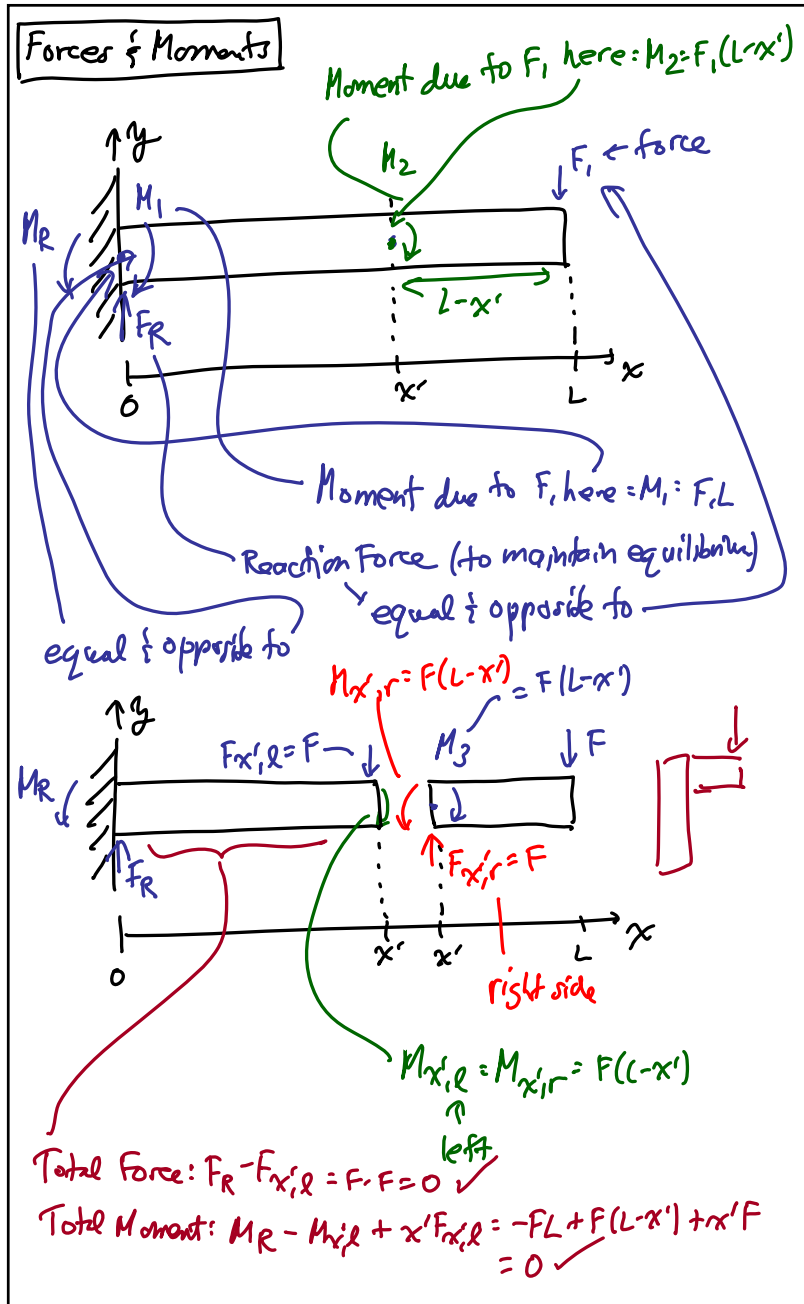


Comb-Driven Folded Beam Actuator

Problem: Bending a Cantilever Beam



- Objective: Find relation between tip deflection $y(x=L_c)$ and applied load F → $F = ky$ ultimately want k
- Assumptions:
 1. Tip deflection is small compared with beam length
 2. Plane sections (normal to beam's axis) remain plane and normal during bending, i.e., "pure bending"
 3. Shear stresses are negligible



Beam Segment in Pure Bending

⇒ Consider the segment bounded by the dashed lines defining $d\theta$

At $z=0$: neutral axis → segment length = $dx = R d\theta$ (1)

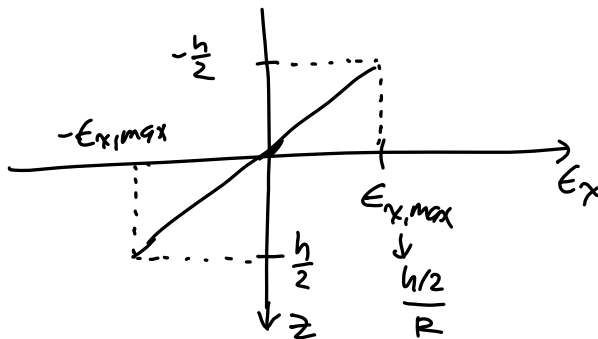
At any z : segment length = $dL = (R-z) d\theta$ (2)

Combine (1) & (2): $dL = dx - z d\theta = dx - \frac{z}{R} dx$

Thus, the axial strain @ z :

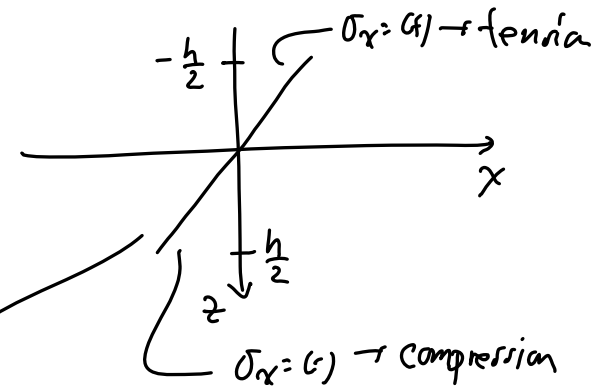
$$\epsilon_x = \frac{dL - dx}{dx} = -\frac{z}{R} \Rightarrow \boxed{\epsilon_x = -\frac{z}{R}}$$

Thus, the strain varies linearly along beam thickness:



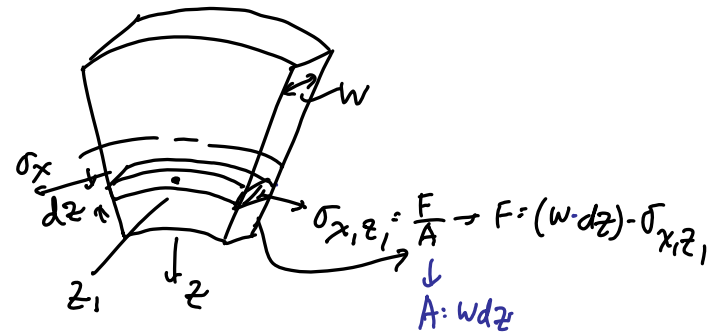
Of course, there is a correspondingly axial stress:

$$\sigma_x = \epsilon_x E = \boxed{-\frac{zE}{R} = \sigma_x}$$



This gradient of stress generates a bending moment (in response to original applied normal force → moment)

Stress → Force:



⇒ integrate stress through the thickness of the beam

differential ~~area~~ moments

moment arm

$$M = \int_{-\frac{h}{2}}^{\frac{h}{2}} \underbrace{[(wdz)\sigma_x]}_{\text{force}} \cdot z$$

$$= \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{EWz^2}{R} dz \Rightarrow M = -\left(\frac{1}{12}Wh^3\right) \frac{E}{R}$$

↑ $\left[\sigma_x = -\frac{zE}{R}\right]$

↓ $\frac{1}{12}Wh^3 = I \triangleq \text{Moment of Inertia}$

Note: (+) radius of curvature
↓
(-) internal bending moment

$$\frac{1}{R} = -\frac{M}{EI}$$

Differential Equation for Beam Bending

Write out some geometric relationships:

⇒ then use small angle approx:

$$\cos\theta = \frac{dx}{ds} \rightarrow ds = \frac{dx}{\cos\theta} \rightarrow ds \cong dx$$

$$\tan\theta = \frac{dw}{dx} = \text{slope of the beam at the point of interest} \rightarrow \theta \cong \frac{dw}{dx} \quad (1)$$

$$ds = R d\theta \rightarrow \frac{1}{R} = \frac{d\theta}{ds} \rightarrow \frac{1}{R} = \frac{d^2w}{dx^2} \quad (2)$$

Inverting (1) into (2):

$$\frac{1}{R} = \frac{d^2w}{dx^2} = -\frac{M}{EI} \leftarrow \text{Diff. Eq. for Small Angle Beam Bending}$$

Cantilever Beam w/ Concentrated Load

Internal Moment

Free end condition

Clamped end condition:
At $x=0$:
 $y=0$
 $dy/dx = 0$

Internal Moment @ position x : $M = -F(L-x)$

Thus: $\frac{d^2 w}{dx^2} = \frac{F}{EI}(L-x)$

w/ $\left\{ \begin{array}{l} \text{Clamped End B.C.: } w(x=0) = 0, \frac{dw}{dx}(x=0) = 0 \\ \text{Free End:} \end{array} \right.$

Solve to get w :

\Rightarrow use Laplace; or a trivial solution:

$w = A + Bx + Cx^2 + Dx^3$, then apply B.C.s

$$w = \frac{FL}{2EI} x^2 \left(1 - \frac{x}{3L}\right)$$

Deflection @ x due to a point load F applied @ $x=L$.

Maximum Deflection \rightarrow occur $x=L$:

$$w_{max} = \left(\frac{L^3}{3EI}\right)F \rightarrow F = \underbrace{\left(\frac{3EI}{L^3}\right)w(x=L)}_{= k_c w(x=L)}$$

where $k_c = \frac{3EI}{L^3}$ stiffness @ $x=L$

$[I = \frac{1}{12}wh^3] \Rightarrow k_c = \frac{1}{4}EW\frac{h^3}{L^3}$

Ex: $L = 100\mu\text{m}, W = 2\mu\text{m}, h = 2\mu\text{m}$

polysilicon $\rightarrow E = 150 \text{ GPa}$

$$k_c = \frac{1}{4}(150\text{G})(2\mu)\left(\frac{2\mu}{100\mu}\right)^3 = \underline{\underline{0.6 \text{ N/m}}}$$

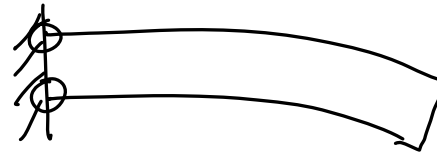
Maximum Strain in a Bent Cantilever

From before, the radius of curvature is given by

$$\frac{1}{R} = \frac{d^2 w}{dx^2} = \frac{F}{EI}(L-x)$$

$\Rightarrow \frac{1}{R}$ is maximized (i.e., R is minimized) when

$$x=0 \quad [x=0] \quad \frac{1}{R} = \frac{d^2 w}{dx^2} = \frac{FL}{EI}$$



Strain is maximized:

- ① At the top surface \rightarrow tensile
- ② At the bottom surface \rightarrow compressive

$$\epsilon_{max} = \frac{z}{R} = \frac{h}{2R} = \frac{h}{2} \frac{FL}{EI} = \epsilon_{max}$$