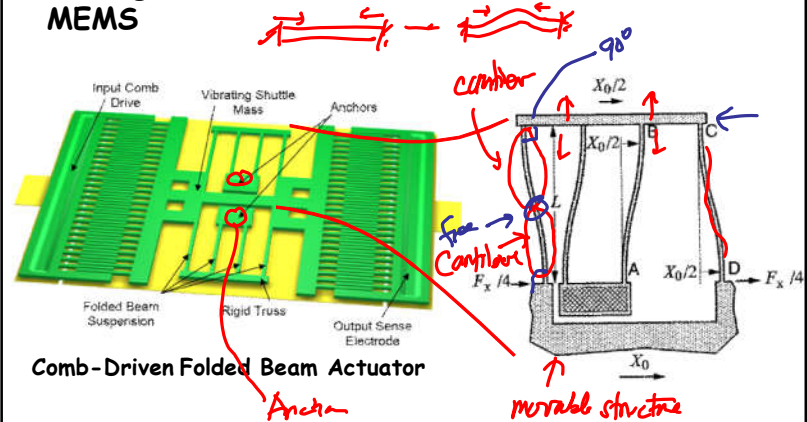


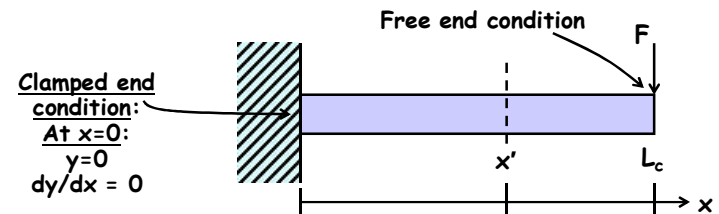
**Lecture 13: Beam Bending**

- Announcements:
- HW#4 online soon
- Module 8 on "Microstructural Elements" online
- Graded HW#1 handed back
- -----
- Reading: Senturia, Chpt. 9
- Lecture Topics:
  - ↪ Bending of beams
  - ↪ Cantilever beam under small deflections
  - ↪ Combining cantilevers in series and parallel
  - ↪ Folded suspensions
  - ↪ Design implications of residual stress and stress gradients
- -----
- Last Time:
- Started Bending of Beams
- Continue with this ...

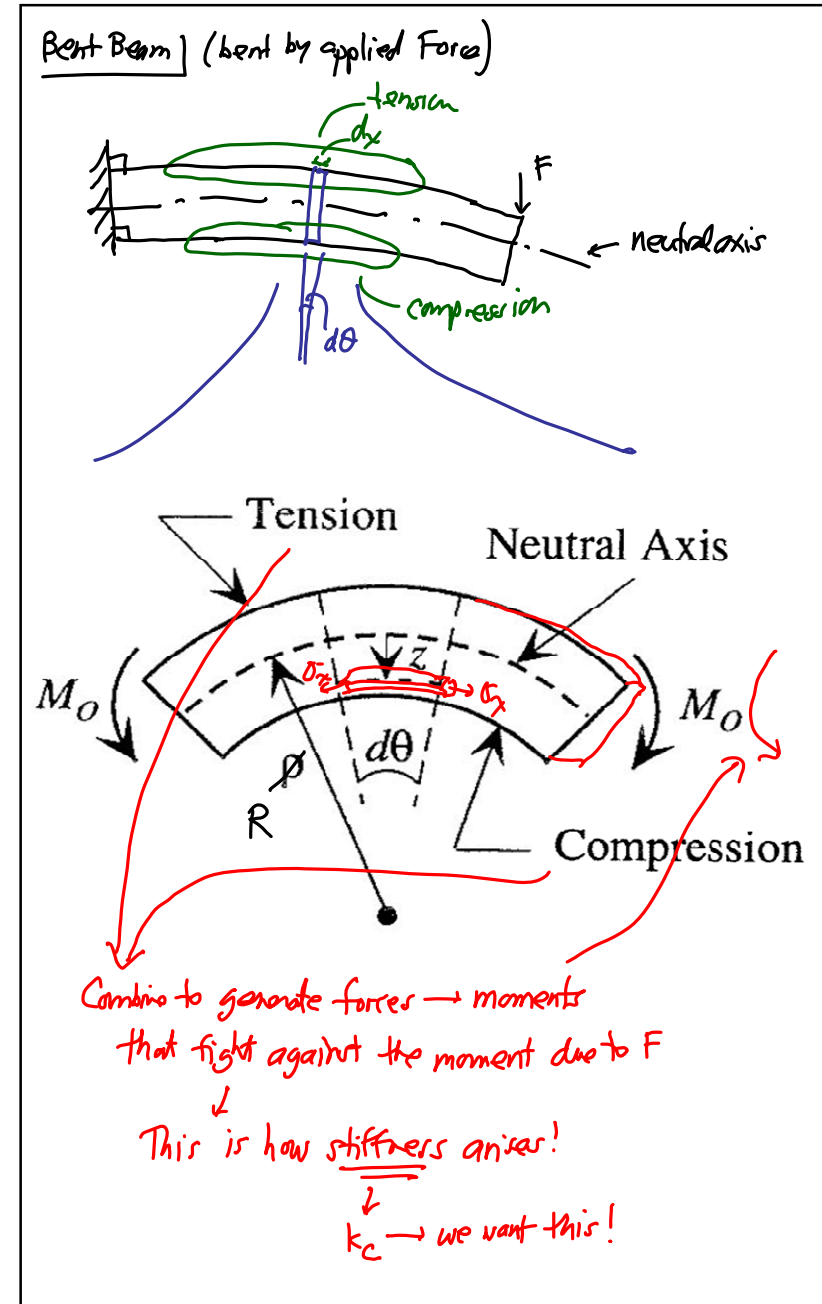
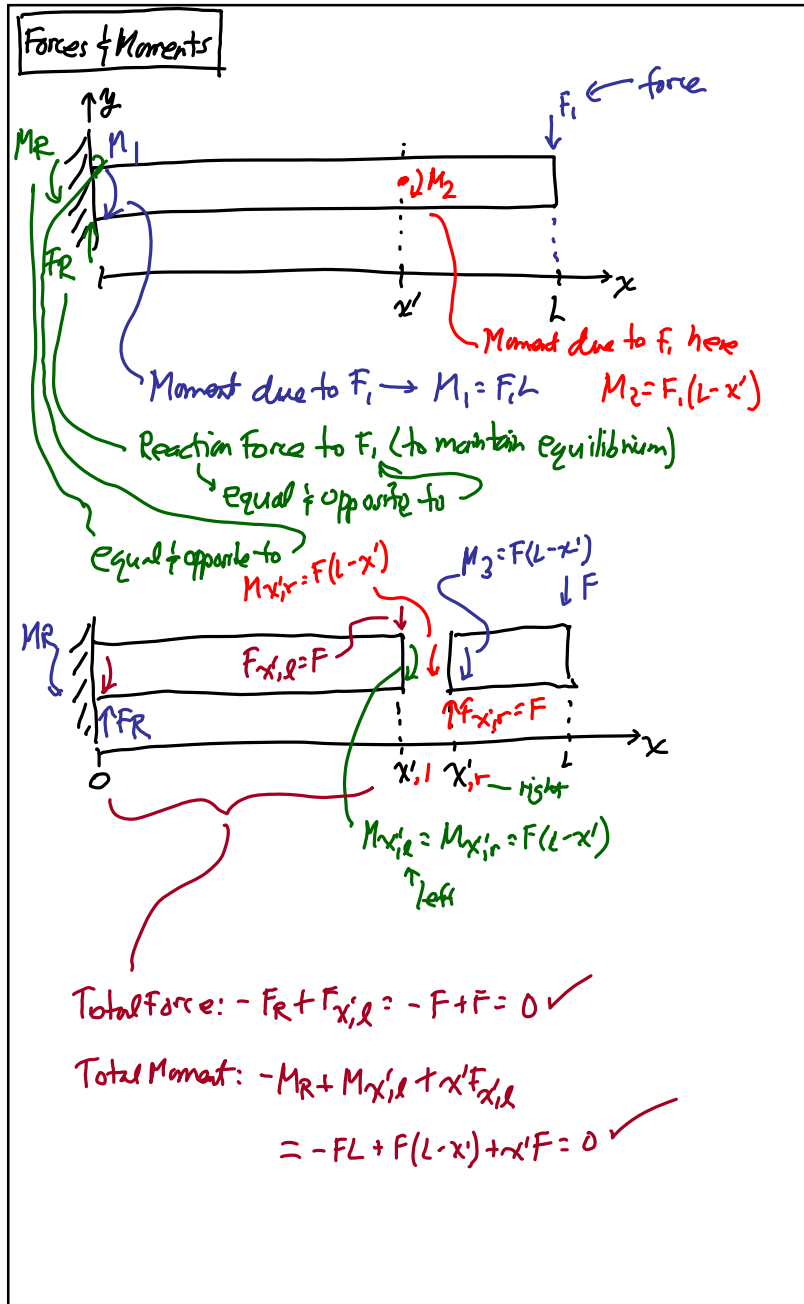
- Springs and suspensions very common in MEMS
- Coils are popular in the macro-world; but not easy to make in the micro-world
- Beams: simpler to fabricate and analyze; become "stronger" on the micro-scale → use beams for MEMS



**Problem: Bending a Cantilever Beam**



- Objective: Find relation between tip deflection  $y(x=L_c)$  and applied load  $F$
- Assumptions:
  1. Tip deflection is small compared with beam length
  2. Plane sections (normal to beam's axis) remain plane and normal during bending, i.e., "pure bending"
  3. Shear stresses are negligible



Beam Segment in Pure Bending

⇒ consider the segment bounded by the dashed lines defining  $d\theta$

At  $z=0$ : neutral axis → segment length =  $dx = R d\theta$  (1)

At any  $z$ : segment length =  $dL = (R-z) d\theta$  (2)

Combine (1) & (2):  $dL = dx - z d\theta = dx - \frac{z}{R} dx$

Thus, the axial strain @  $z$ :  $\epsilon_x = \frac{dL - dx}{dx} = -\frac{z}{R}$

$$\epsilon_x = -\frac{z}{R}$$

Thus, the strain varies linearly along the beam thickness!

Of course, there is a corresponding axial stress:

$$\sigma_x = \epsilon_x E = \boxed{-\frac{zE}{R} = \sigma_x}$$

This gradient of stress then generates a bending moment!  
in response to the original applied moment (from F)

Stress → Force:

$M = Fz$

$F = A \sigma_x = (wdz) \sigma_x$

$A = wdz$

⇒ integrate stress through the beam thickness:

$$M = \int_{-\frac{h}{2}}^{\frac{h}{2}} \underbrace{[Wdz] \sigma_x}_{\text{force}} \cdot z$$

$$\Rightarrow \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{EWz^2}{R} dz \Rightarrow M = -\left(\frac{1}{12}Wh^3\right) \frac{E}{R}$$

$\left[\sigma_x = -\frac{zE}{R}\right]$

$\frac{1}{12}Wh^3 = I \triangleq$  Moment of Inertia

Note: (+) radius of curvature  
(-) internal bending moment

$$\frac{1}{R} = -\frac{M}{EI}$$

Differential Equation for Beam Bending

Write out some geometric relationships:

⇒ then use small angle approximation

$$\cos\theta = \frac{dx}{ds} \rightarrow ds = \frac{dx}{\cos\theta} \rightarrow ds = dx$$

$$\tan\theta = \frac{dw}{dx} = \text{slope of the beam at this point} \rightarrow \theta \approx \frac{dw}{dx} \quad (1)$$

$$ds = R d\theta \rightarrow \frac{1}{R} = \frac{d\theta}{ds} \rightarrow \frac{1}{R} = \frac{d^2w}{dx^2} \quad (2)$$

Inserting (1) into (2):

$$\frac{1}{R} = \frac{d^2w}{dx^2} = -\frac{M}{EI}$$

Diff. Eq. for Small Angle Beam Bending

Cantilever Beam w/ Concentrated Load

Clamped end condition:  
At  $x=0$ :  
 $y=0$   
 $dy/dx = 0$

Free end condition

Point Load  $F$

Internal Moment

Internal Moment @ position  $x$ :  $M = -F(L-x)$

Thus:  $\frac{d^2w}{dx^2} = \frac{F}{EI}(L-x)$

$w(x)$  { Clamped End B.C.:  $w(x=0) = 0, \frac{dw}{dx}(x=0) = 0$   
Free End B.C.: none

Solve to get  $w$ :

⇒ use Laplace; or a trial solution

$w = A + Bx + Cx^2 + Dx^3$ , then apply B.C.'s

$$w = \frac{FL}{2EI} x^2 \left(1 - \frac{x}{3L}\right)$$

Deflection @  $x$  due to a point load  $F$  applied @  $x=L$

Maximum Deflection → occurs  $x=L$ :

$$w_{max} = \left(\frac{L^3}{3EI}\right) F \rightarrow F = \left(\frac{3EI}{L^3}\right) w(x=L) = k_c w(x=L)$$

Stiffness  $\hat{=} k_c$  @  $x=L$

$$k_c = \frac{3EI}{L^3} \rightarrow k_c = \frac{1}{4} E W \frac{h^3}{L^3}$$

$$[I = \frac{1}{12} W h^3]$$

Ex:  $L = 100 \mu\text{m}, W = 2 \mu\text{m}, h = 2 \mu\text{m}$

polysilicon →  $E = 150 \text{ GPa}$

$$k_c = \frac{1}{4} (150 \text{ G}) (2 \mu) \left(\frac{2 \mu}{100 \mu}\right)^3 = \underline{\underline{0.6 \text{ N/m}}}$$

Maximum Stress in a Bent Cantilever

From before, the radius of curvature is given by:

$$\frac{1}{R} = \frac{d^2 w}{dx^2} = \frac{F}{EI} (L-x)$$

⇒  $\frac{1}{R}$  is maximized (i.e.,  $R$  is minimized) when

$$x=0: [x=0] \rightarrow \frac{1}{R} = \frac{d^2 w}{dx^2} = \frac{FL}{EI}$$



Strain is maximized

- ① At top surface → tensile
  - ② At bottom surface → compressive
- } @  $x=0$

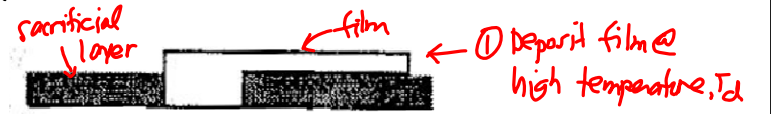
$$\epsilon_{max} = \frac{z}{R} = \frac{h}{2} \frac{1}{R} = \frac{h}{2} \frac{FL}{EI} = \epsilon_{max}$$

$$[I = \frac{1}{12} W h^3] \rightarrow \epsilon_{max} = \frac{1}{2} \frac{FL}{E} \left(\frac{12}{W h^3}\right) = \frac{6L}{E W h^2} F$$

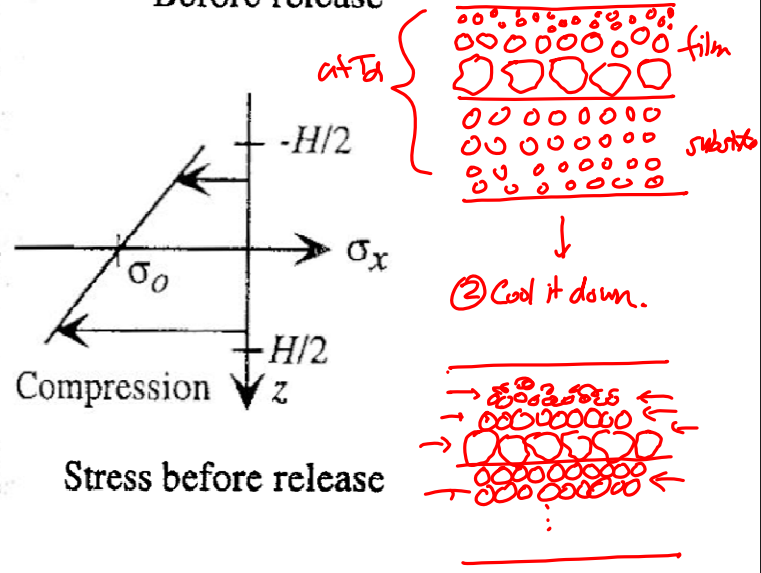
$$\sigma_{max} = \epsilon_{max} E = \frac{6L}{wh^2} F$$

(Maximum stress in a bent cantilever  
 subjected to a force F at its tip)

Stress Gradient in Cantilevers



Before release



Stress before release