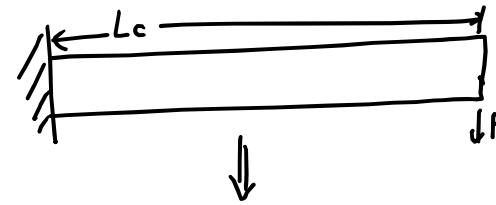


Lecture 17: Resonance Frequency

- **Announcements:**
- This is a video lecture, since I am at the EECS Faculty Retreat this day
- Module 9 on Energy Methods online
- Module 10 on Resonance Frequency online
- Midterm Exam: Thursday, March 21, 11-12:30 p.m., 293 Cory (right here)
- 
- Reading: Senturia, Chpt. 10
- Lecture Topics:
  - ↳ Energy Methods
  - ↳ Virtual Work
  - ↳ Energy Formulations
  - ↳ Tapered Beam Example
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- Reading: Senturia, Chpt. 10: §10.5, Chpt. 19
- Lecture Topics:
  - ↳ Estimating Resonance Frequency
  - ↳ Lumped Mass-Spring Approximation
  - ↳ ADXL-50 Resonance Frequency
  - ↳ Distributed Mass & Stiffness
  - ↳ Folded-Beam Resonator
  - ↳ Resonance Frequency Via Differential Equations
- 
- Last Time:
- Working through energy methods
- Continue with this ...

Same Problem as Before: Take a beam, apply a force.



① Apply force.



② Beam responds by bending.

④ Strain generated.

So the beam has received an influx of stored energy.

③ This force has done work:

$$W = F \cdot y(L_c)$$

magnitude of " " determined by shape.

⑤ Then

$$U = \text{Stored Energy} - \text{Work Done} \rightarrow 0$$

transfer function  $y(x) = f(x)$  (When we choose the right shape.)

This is how we get the beam's response to  $F$ !

Fundamentals: Energy Density

General Definition of Work:

$$W(q_1) = \int_0^{q_1} e(q) dq \quad \begin{array}{l} q: \text{displacement} \\ e: \text{effort} \end{array}$$

for EE:  $W(Q) = \int_0^Q \frac{Q}{C} dQ$

Strain Energy Density

$$w = \int_0^{\epsilon_x} \sigma_x d\epsilon_x \quad \leftarrow \text{value of strain @ position } (x, y, z)$$

$\sigma_x(\epsilon_x) \rightarrow$  relates stress to strain @ position  $(x, y, z)$

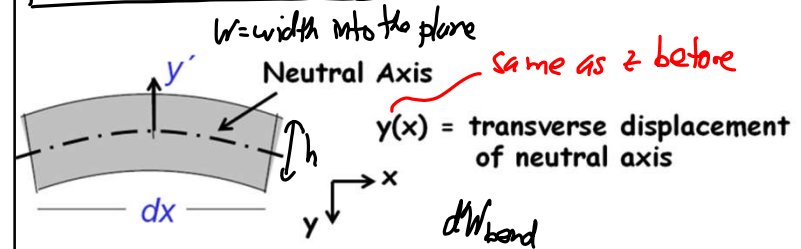
$[\sigma_x = E\epsilon_x]$

$$w = \int_0^{\epsilon_x} E\epsilon_x d\epsilon_x = \frac{1}{2} E\epsilon_x^2$$

Total Strain Energy: [J]

$$W = \iiint \left( \frac{1}{2} E(\epsilon_x^2 + \epsilon_y^2 + \epsilon_z^2) + \frac{1}{2} G(\gamma_{xy}^2 + \gamma_{xz}^2 + \gamma_{yz}^2) \right) dV \quad \leftarrow \text{Volume}$$

Bending Energy Density



First, find the bending energy  $\wedge$  in an infinitesimal length  $dx$

$$dW_{\text{bend}} = w dx \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{1}{2} E \epsilon_x^2(y') dy'$$

$$\left[ \frac{1}{R} = \frac{d^2 y}{dx^2}, \epsilon_x = \frac{y'}{R} \right] \Rightarrow \epsilon_x(y) = y' \frac{d^2 y}{dx^2}$$

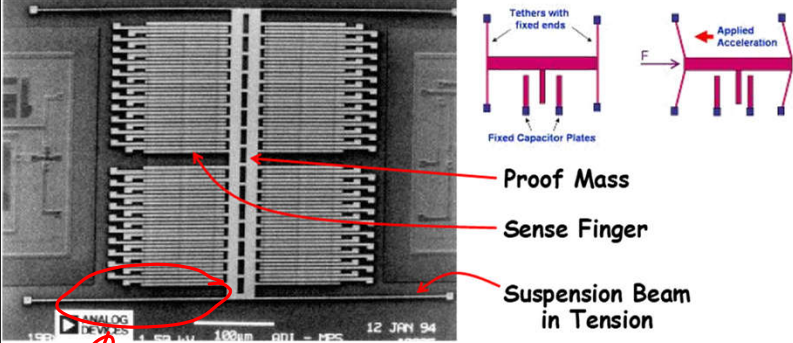
$$\begin{aligned} dW_{\text{bend}} &= w dx \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{1}{2} E \left[ y' \frac{d^2 y}{dx^2} \right]^2 dy' \\ &= \frac{1}{2} E \left( \frac{wh^3}{12} \right) \left( \frac{d^2 y}{dx^2} \right)^2 dx \\ &\quad \underbrace{\hspace{1.5cm}}_{I_z} \end{aligned}$$

$$\therefore W_{\text{bend}} = \frac{1}{2} E I_z \int_0^L \left( \frac{d^2 y}{dx^2} \right)^2 dx$$

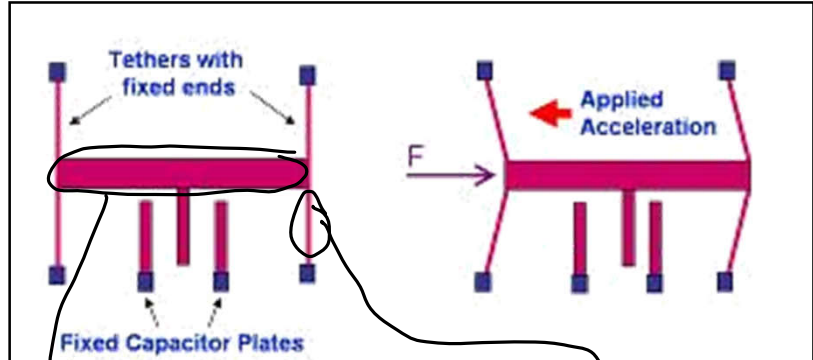


\*  $\omega_0 = \sqrt{\frac{k}{m}}$   $\Rightarrow$  good for problems where mass & stiffness can be separated  
i.e., they are distinct

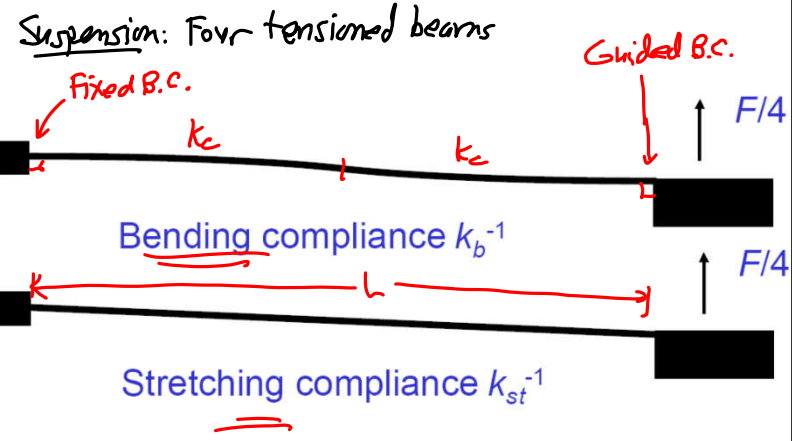
- The proof mass of the ADXL-50 is many times larger than the effective mass of its suspension beams
  - Can ignore the mass of the suspension beams (which greatly simplifies the analysis)
- Suspension Beam:  $L = 260 \mu\text{m}$ ,  $h = 2.3 \mu\text{m}$ ,  $W = 2 \mu\text{m}$



In fabrication: purposely introduce a tensile stress in the beams!  
a large one  $\rightarrow$  why?  
 $\rightarrow$  to avoid compression at all cost  
 $\rightarrow$  buckling  $\rightarrow$  dead device



mass of structure  $\gg$  mass of the springs  
 $\therefore$  ignore the mass of the springs  
stiffness of the springs  $\ll$  stiffness of structure  
 $\therefore$  ignore the stiffness of the structure  
for the ADXL-50, 60% of the mass comes from the sense fingers  $\rightarrow M = 162 \text{ ng}$

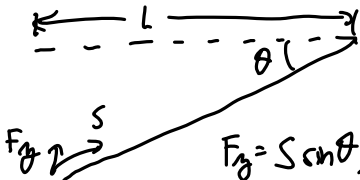


Bending Contribution

$$k_b = k_{cl} k_c = \left(\frac{1}{k_c} + \frac{1}{k_c}\right)^{-1} = \frac{k_c}{2} = \frac{1}{2} \frac{3E(WH^3/12)}{L(1/2)^3}$$

$$\Rightarrow k_b = EW \left(\frac{H}{L}\right)^3 = 0.24 \text{ N/m}$$

Stretching Contribution



$$F_g = S \sin \theta \approx \frac{y}{L} S \left(\frac{y}{L}\right) = \left(\frac{S}{L}\right) y$$

(assume small displacement)  $k_{st}$  stretching stiffness

$$k_{st} = \frac{S}{L} = \frac{0.1 \text{ N}}{L} = 0.88 \text{ N/m}$$

Get the total spring constant

bending stiffness } parallel → add!  
stretching stiffness }

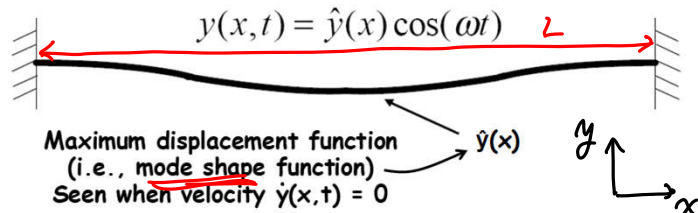
$$k_{tot} = 4(k_b + k_{st}) = 4(0.24 + 0.88) = 4.5 \text{ N/m}$$

Now, get the resonance freq:

$$f_0 = \frac{1}{2\pi} \sqrt{\frac{k}{m}} = \frac{1}{2\pi} \sqrt{\frac{4.48 \text{ N/m}}{162 \times 10^{-12} \text{ kg}}} = 26.5 \text{ kHz}$$

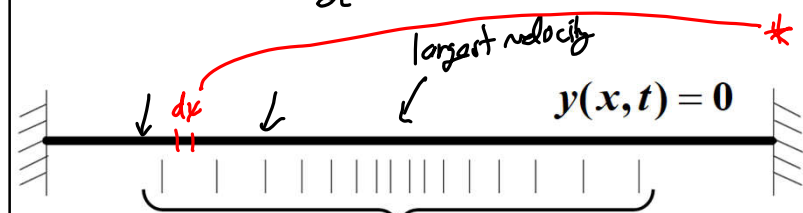
APXL-30 DataSheet:  $f_0 = 24 \text{ kHz}$  ← difference?  
→ Capacitive transducer → electrical stiffness

Find the Resonance Frequency when Mass & Stiffness are Distributed

- Vibrating structure displacement function:
 
- Procedure for determining resonance frequency:
  - Use the static displacement of the structure as a trial function and find the strain energy  $W_{max}$  at the point of maximum displacement (e.g., when  $t=0, \pi/\omega, \dots$ )
  - Determine the maximum kinetic energy when the beam is at zero displacement (e.g., when it experiences its maximum velocity)
  - Equate energies and solve for frequency

Get Maximum Kinetic Energy

Velocity:  $v(x,t) = \frac{\partial y(x,t)}{\partial t} = -\omega \hat{y}(x) \sin \omega t$



Velocity topographical mapping

When  $y(x,t) = 0$ , all the energy in the structure is kinetic ( $W = 0$ ) ←  $t = \frac{\pi}{2\omega}, \frac{3\pi}{2\omega}, \dots$

$$v(x, \frac{(2m+1)\pi}{2\omega}) = -\omega \hat{y}(x)$$

$W$   
 $h$   
 $dx$   
 $v$   
 $\rho$  density  
 $dK = \frac{1}{2} \cdot dm \cdot [v(x,t)]^2$   
 $dm = \rho(W h dx)$   
 $\hat{\eta}(x)$  resonance mode shape

Maximum Kinetic Energy,  $K_{max}$ :  
 $K_{max} = \int_0^L \frac{1}{2} \rho W h dx v^2(x,t) = \int_0^L \frac{1}{2} \rho W h \omega^2 [\hat{\eta}(x)]^2 dx$   
 To get frequency:  $K_{max} = W_{max}$

$\therefore \omega = \sqrt{\frac{W_{max}}{\int_0^L \frac{1}{2} \rho W h [\hat{\eta}(x)]^2 dx}} \quad [\text{radians/s}]$   
 $\omega$ : radian resonance freq.  
 $W_{max}$ : maximum potential energy  
 $\rho$ : density of the structural material  
 $W$ : beam width  
 $h$ : " thickness  
 $\hat{\eta}(x)$ : resonance mode shape

### Resonance Freq. of a Folded Beam Structure

Folded-beam suspension  
 Shuttle w/ mass  $M_s$   
 Folding truss w/ mass  $M_t$   
 Anchor  $h = \text{thickness}$

- Derive an expression for the resonance frequency of the above structure

Approximation  
 $\Rightarrow m = \text{shuttle mass}$   
 $\Rightarrow k = k_c$

$\omega_0 = \sqrt{\frac{k_c}{m}}$   
 But not accurate enough for some applications.  
 $\Rightarrow$  for better accuracy, must integrate



Use the Rayleigh-Ritz Method: (energy method)

$$\mathcal{K}_{\max} = \mathcal{W}_{\max} \leftarrow \frac{1}{2} k x^2$$

Find the kinetic energy  $\rightarrow$  one piece at a time

$$\mathcal{K}_{\max} = \underbrace{\mathcal{K}_s}_{\text{shuttle}} + \underbrace{\mathcal{K}_t}_{\text{truss}} + \underbrace{\mathcal{K}_b}_{\text{beams}}$$

$$= \frac{1}{2} v_s^2 M_s + \frac{1}{2} v_t^2 M_t + \frac{1}{2} \int v_b^2 dm_b$$

Velocity of the Shuttle:  $v_s = \omega_0 x_0$   
 $\uparrow$   $\leftarrow$  maximum displacement  
 res. freq.  $\uparrow$  of shuttle

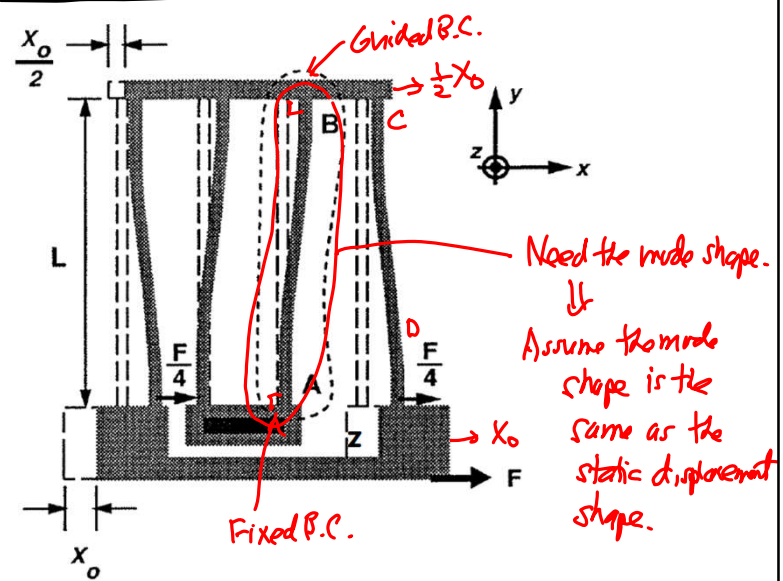
$$\therefore \mathcal{K}_s = \frac{1}{2} v_s^2 M_s = \left( \frac{1}{2} \omega_0^2 x_0^2 M_s \right) = \mathcal{K}_s$$

Velocity of Truss:  $v_t = \frac{1}{2} v_s = \frac{1}{2} \omega_0 x_0$

$$\therefore \mathcal{K}_t = \frac{1}{2} \left( \frac{1}{2} \omega_0 x_0 \right)^2 M_t = \left( \frac{1}{8} \omega_0^2 x_0^2 M_t \right) = \mathcal{K}_t$$

$\uparrow$   
mass of both trusses

Velocity of the Beam Segments:  $\rightarrow$  first beam [AB]



Segment [AB]:

$$\hat{x}(\eta) = \frac{F x}{48 E I_2} (3\eta^2 - 2\eta^3), \quad 0 \leq \eta \leq L \quad (1)$$

$$\text{At } \eta = L: \quad x(L) = \frac{x_0}{2} = \frac{F x L^3}{48 E I_2} \leftarrow \text{B.C.}$$

Substitute into (1):

$$\hat{x}(\eta) = \frac{x_0}{2} \left[ 3 \left( \frac{\eta}{L} \right)^2 - 2 \left( \frac{\eta}{L} \right)^3 \right]$$

which yields for velocity:

$$v_b(\eta)|_{[AB]} = \frac{x_0}{2} \left[ 3 \left( \frac{\eta}{L} \right)^2 - 2 \left( \frac{\eta}{L} \right)^3 \right] \omega_0$$

Plugging into expression for  $\mathcal{K}_b$ :

$$\mathcal{K}_{[AB]} = \frac{1}{2} \int_0^L \frac{X_0^2 \omega_0^2}{4} \left[ 3 \left( \frac{y}{L} \right)^2 - 2 \left( \frac{y}{L} \right)^3 \right]^2 dM_{[AB]}$$

$$= \frac{X_0^2 \omega_0^2 M_{[AB]}}{8L} \int_0^L \left[ 3 \left( \frac{y}{L} \right)^2 - 2 \left( \frac{y}{L} \right)^3 \right]^2 dy$$

mass per unit length

$M_{[AB]}$  = static mass

$$\mathcal{K}_{[AB]} = \frac{13}{2880} X_0^2 \omega_0^2 M_{[AB]}$$

For segment [CD]:

$$v_s(y)|_{[CD]} = X_0 \left[ 1 - \frac{3}{2} \left( \frac{y}{L} \right)^2 + \left( \frac{y}{L} \right)^3 \right] \omega_0$$

Thus:

$$\mathcal{K}_{[CD]} = \frac{X_0^2 \omega_0^2 M_{[CD]}}{2L} \int_0^L \left[ 1 - \frac{3}{2} \left( \frac{y}{L} \right)^2 + \left( \frac{y}{L} \right)^3 \right]^2 dy$$

$$\mathcal{K}_{[CD]} = \frac{83}{2880} X_0^2 \omega_0^2 M_{[CD]}$$

Let  $M_b \triangleq$  total mass of all  
& beams

static mass of  
beam [CD]

Thus:

$$\mathcal{K}_b = 4\mathcal{K}_{[AB]} + 4\mathcal{K}_{[CD]} = \frac{6}{35} X_0^2 \omega_0^2 M_b$$

and

$$\mathcal{K}_{\max} = X_0^2 \omega_0^2 \left[ \frac{1}{2} M_s + \frac{1}{8} M_t + \frac{6}{35} M_b \right]$$

← for the total mechanical clt.

↑ both trusses    ↑ all beams

$\mathcal{W}_{\max}$  → max. potential energy → equal to the work done to achieve maximum deflection

$$\mathcal{W}_{\max} = \frac{1}{2} k_x X_0^2$$

Then, using Rayleigh-Ritz:

$$\mathcal{K}_{\max} = \mathcal{W}_{\max}$$

$$X_0^2 \omega_0^2 \left[ \frac{1}{2} M_s + \frac{1}{8} M_t + \frac{6}{35} M_b \right] = \frac{1}{2} k_x X_0^2$$

$$\omega_0 = \left[ \frac{k_c}{M_{eq}} \right]^{1/2}$$

$$\text{where } M_{eq} = M_s + \frac{1}{4} M_t + \frac{12}{35} M_b$$

(Resonance Freq. of a Folded-Beam  
Suspended Shuttle)